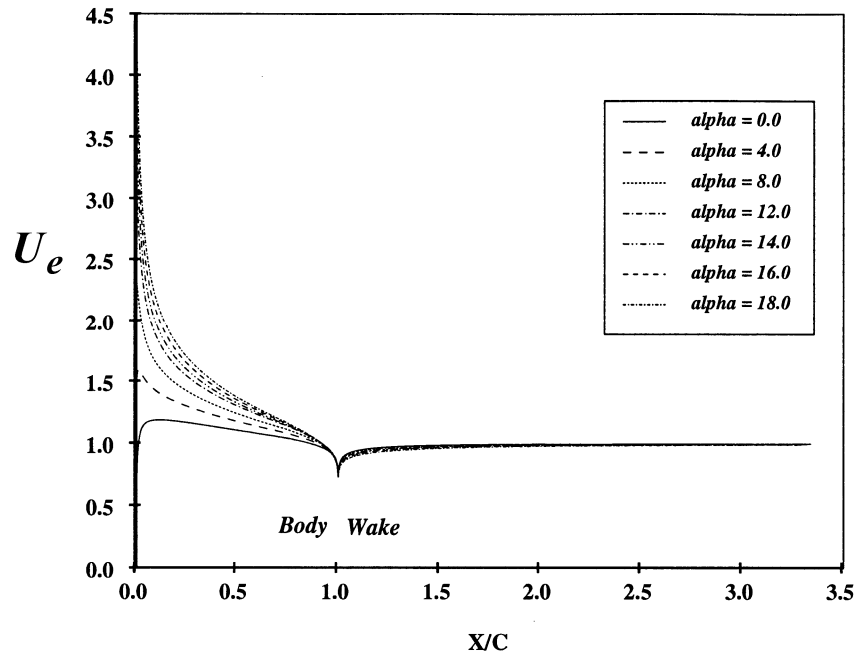
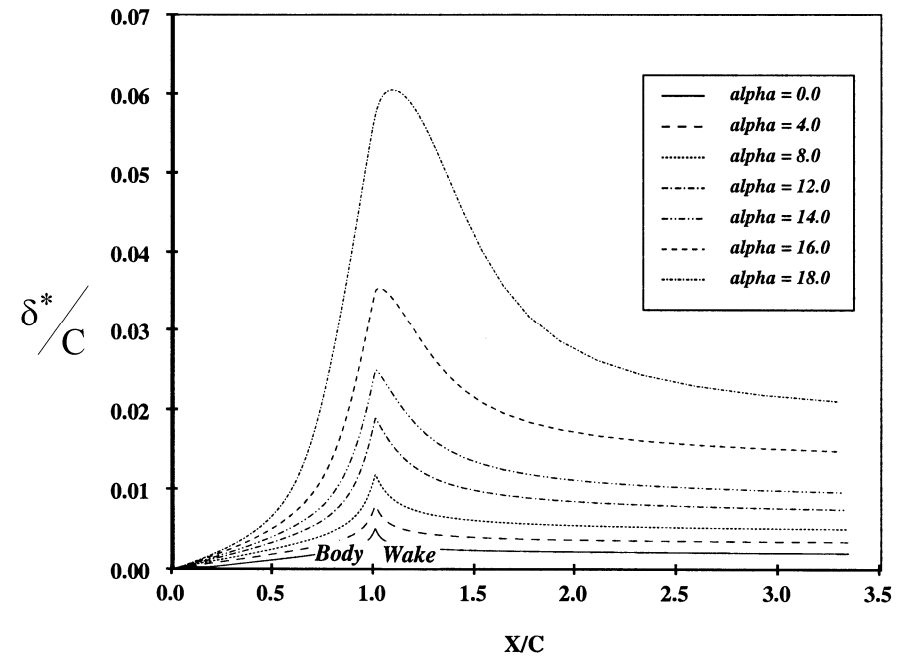


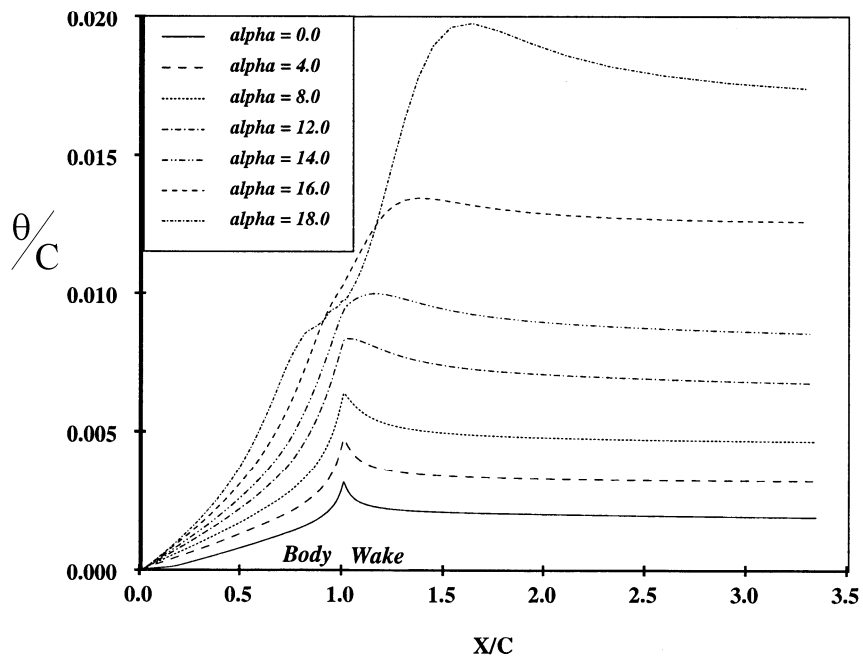
Inviscid U_e Distribution
N0012 Airfoil



Displacement Thickness Distribution
N0012 Airfoil at $Re = 4.0$ M



Momentum Thickness Distribution
N0012 Airfoil at $Re = 4.0$ M



C_f Distribution
N0012 Airfoil at $Re = 4.0$ M

